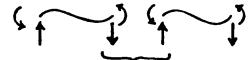
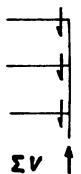
COLUMNS: SPECIAL TRANSVERSE REINFORCEMENT-CONTO

SPECIAL TRANSV. REINF. IS ALSO REQUIRED WHERE COLUMN CAPACITY IS LESS THAN THE SUM OF THE SHEARS ABOVE. REF. APP. C. PARAGRAPH C/2



- . IV IS THE COLUMN LOAD AT THIS LEVEL.
- · AT INTERIOR COL'S THIS SUM IS RELATIVELY SMALL.
- · END COLUMNS ARE USUALLY CRITICAL.



I. INCLUDE ALL BEAMS ABOVE THE COLUMN IN QUESTION.

3. AT THE COLUMN IN QUESTION, CALCULATE THE MAX. MOMENT TRANSFERRED TO THE COLUMN BY THE YIELDING BEAM.

4. DOES THE COLUMN HAVE THE CAPACITY TO CARRY EV, WITH THIS BEAM MOMENT?
YES: NO ADD'L REINF, REQ'D.
NO: PROVIDE THE SPECIAL TRANSV. REINF, CALCULATED ON P.19

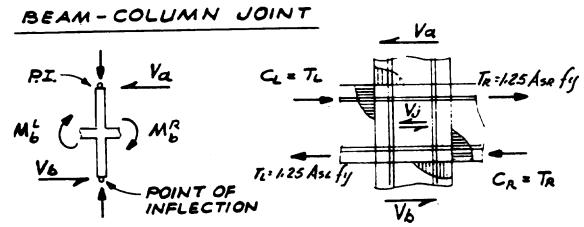
FOR FULL HEIGHT

see p. 21 for sample calc.

COLUMNS: SPECIAL TRANSVERSE REINFORCEMENT CONT.

FRAME	ß	
COLUMN	1	
ROOF BEAM		
EMp/L	35	
VotL	55	
Va	90	CALCS, NOT SHOWN
3RD FLR. BEAM		
EMp/L	43	
VD+L	67	
V _u ³	110	ASSUME SAME AS 2ND
ZND FLR. BEAM		
ZMp/L	43	= 794 + 489 (p. 18)
VOLL	67	$= \frac{794 + 489}{(p.12)^{30}} (p.13)$
V.5	110	
E Vu ABOVE	310	
Mp from BM	397	$=\frac{1}{2}BMMp=\frac{794}{2}$
ALLOW COL. M WITH P= E Vu	659	(ACI. SP 17A VOLZ CHARTE4-60-75)
COL. M > Mp	YES	
SPEC. TRANSV. REINF.	NO	

Figure D-2. Continued.



FORCES ON COLUMN

HOOP REINF.

FORCES ON BEAM COLUMN JOINT

THE JOINT SHEAR, Vj = 1.25 AsR fy +C1-Va = 1.25 (Asa + Asi) fy - Va Vj = Vj / Ø Aj

Aj = BEAM WINTH x COLUMN DEPTH = 28" x 24"

INTERIOR JOINTS: REF. ACI 21.6.2,3

JOINTS ARE CONFINED $\therefore V_j \leq (20 \sqrt{f_z'} = 1265 psi)$ TRANSV. HOOP REINF. MAY BE $\frac{1}{2}$ OF COLUMN

EXTERIOR JOINTS:

JOINTS ARE NOT CONFINED

.: Vj \leq (15\sqrt{f}' = 949 psi)

TRANSV. HOOPS SAME AS COL. HOOPS

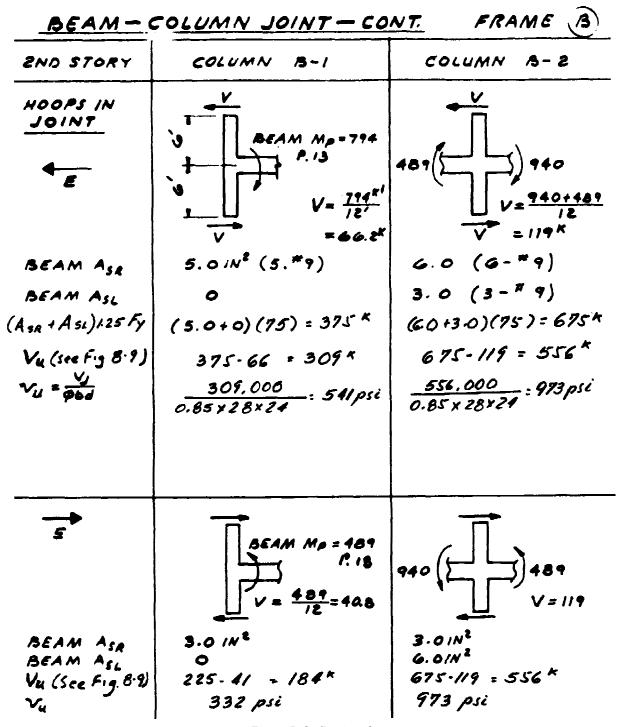
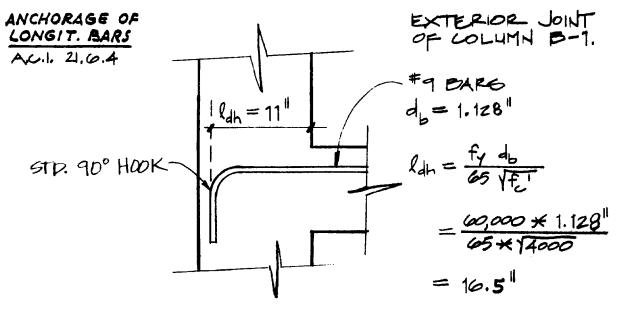
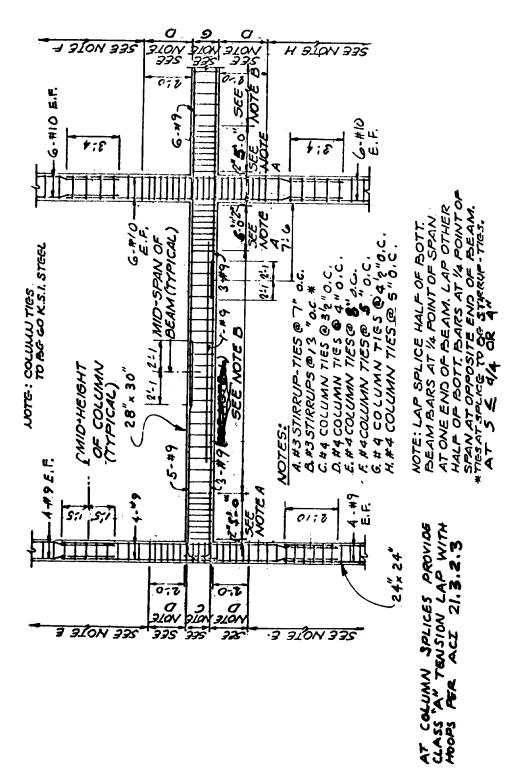


Figure D-2. Continued.

BEAM-CO	LUMN JOINT - CONT.	FRAME B		
2ND STORY	COLUMN B-1	COLUMN B-Z		
Ŋ	541 < 949 p.s.i. o.k.	973 < 1265 p.s.i. o.k.		
#4 H00PS	SAME AS COL. (p. 19)	12 THAT OF COL. (p. 19)		
Hoop SPACING	4"	811		





LONGITUDINAL FRAME - LINE B

DESIGN EXAMPLE D-3

Steel Special Moment Resisting Frame and Steel Braced Frame

Description of Structure. A three-story Administration Building with transverse special moment-resisting frames and longitudinal concentric or eccentric braced frames in structural steel, using non-bearing, non-shear, exterior walls (skin) of flexible insulated metal panels. There are a series of interior vertical load-carrying column and girder bents in addition to the space frame. The structural concept is illustrated on Sheets 2 and 3.

Construction Outline.

Roof:

Built-up 5 ply.

Metal decking with
insulation board.
Suspended ceiling.

2nd & 3rd Floors:

Metal decking with concrete fill.
Asphalt tile.
Suspended ceiling.

1st Floor:

Concrete slab-on-grade.

Exterior Walls:
Non-bearing, non shear,

insulated metal panels.

Partitions:

Non-structural removable drywall.

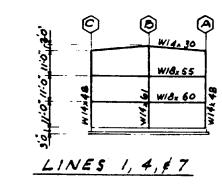
Design Concept. The transverse ductile moment-resisting frames are independent of the logitudinal braced frames. The moment frames are designed to $R_{\rm W}=12$; the concentric braced frames to $R_{\rm W}=8$; the eccentric braced frames to $R_{\rm W}=10$. The metal deck roof system forms a flexible diaphragm; therefore the roof loads are distributed to the frames by tributary area rather than by frame stiffnesses. The metal deck with concrete fill system for the floors form rigid diaphragms and the seismic loads are proportioned to the frames by the frame stiffnesses.

Discussion. Because of the importance of drift of flexible frames, the example shows several stages of design. Preliminary design to find sizes by approximate methods, using different sets of forces for stress and drift. The resulting trial sizes are then used in a computer analysis. (The frames are simple enough to be calculated by hand, but the computer makes short work of calculating design forces, frame period and drift). Final design is discussed, and examples are given for modifications to the results of the computer analysis for accommodating various stress and deflection criteria with consistent sets of member sizes, period, design force, and drift.

Figure D-3. Steel ductile moment resisting space frame and steel braced frame.

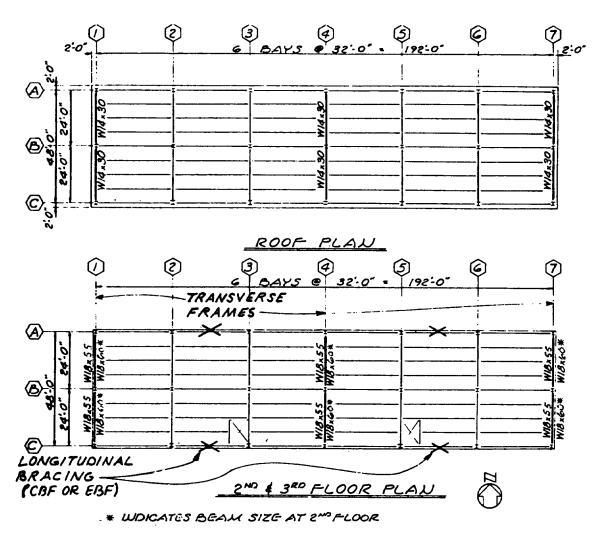
LOADS.

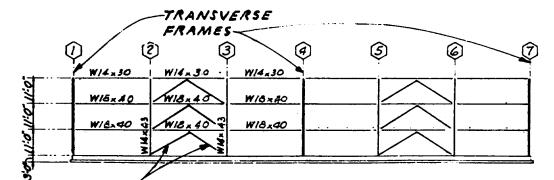
ROOF:			2ND & 3RD FLOORS;		
5-PLY ROOFING	=	6.0 P.S.F.	FINISH	=	I.O P.S.F.
I"INSULATION	=	1.5	STEEL DECK	=	3./
STEEL DECK	=	2. 3	CONCRETE FILL		320
STEEL PURLINS	=	3. 7	STEEL BEAMS	=	5.9
STEEL GIRDERS	=	1.2	STEEL GIRDERS		• •
CEILING	=	10.0	\$ COLUMNS	=	1.5
			PARTITIONS	= .	200
MISCELLANEOUS	=	1. 0	CEILING		100
DEAD LOAD	=	25.7 P.S.F	MISCELLANEOUS	_	1.0
			DEAD LOAD	= =	14.5 P.S.F.
ADD FOR SEISMIC:					,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,
PARTITIONS	= _/	0.0	LIVE LOAD	= ;	50.0 P.S.F.
TOTAL FOR SEISMI	c = 3	15. 7 P.S.F.			



TRANSVERSE SPECIAL MOMENT RESISTING FRAMES
SEE SHT. 7

Figure D-3. Continued.





STEEL TURE

LONGIT. CONCENTRIC BRACED FRAMES-SEE SHT 28
SOUTH ELEVATION (NORTH ELEV. SIM. OPP. HAND)

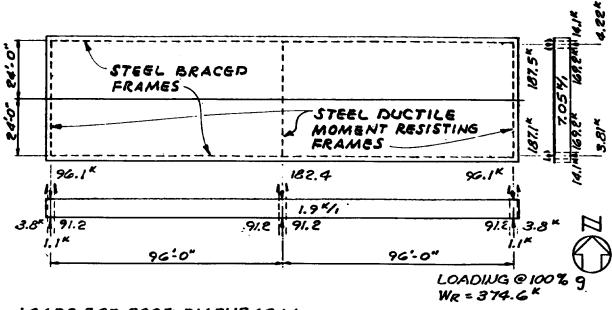
(FOR LONGITUDINAL ECCENTRIC BRACED FRAMES, SEE SHEET 40.)

Figure D-3. Continued.

TM 5-809-10/NAVFAC P-355/AFM 88-3, Chap 13

DESIGN PROCEDURE

	Example Page
A. GENERAL INFORMATION	
 Building Layout Loads for Diaphragms 	1-3 5, 6
B. TRANSVERSE MOMENT RESISTING FRAMES	
 Frame Characteristics Building Period Lateral Forces Distribution of Forces to Frames Preliminary Design Criteria for Final Design Computer Input Computer Output Final Design - Drift Member Stresses Girder-Column Connection Strong Column/Weak Beam 	7 8 10 11 13-16 17 18 19 20 21,22 23 27
C. LONGITUDINAL CONCENTRIC BRACED FRAME 1. Lateral Forces 2. Vertical Forces in Members 3. Seismic Forces in Members 4. Member Sizes 5. Connections 6. Deflection 7. Final Period	28 29 30 31-33 34-36 37-38 39
D. LONGITUDINAL ECCENTRIC BRACED FRAME ALTERNATE	
 Design Procedure Lateral Forces Typical Beam Sizing Preliminary Frame Member Sizing Braced Frame Analysis 	40 41 42 43-45 46
6. Final Member Sizing	47-53



LOADS FOR ROOF DIAPHRAGM

GXTGRIOR WALLS (NON STRUCTURAL GXTGRIOR COVERING)

WALL WY.5.3 PSF x 5.5' = 29.0 %, WALL-WINDOWS OUY

W. WALL = 29 x .75 = 22 x 192' = 4224*

S. WALL = 29 x .68 = 19.8 x 192' = 3801*

41.8 %,

WALL WY.5.3 PSF x 6' = 31.8 %,

G. WALL = 31.8 x .76 = 24 x 48' = 1.152*

W. WALL = 31.8 x .76 = 24 x 48' = 1.152*

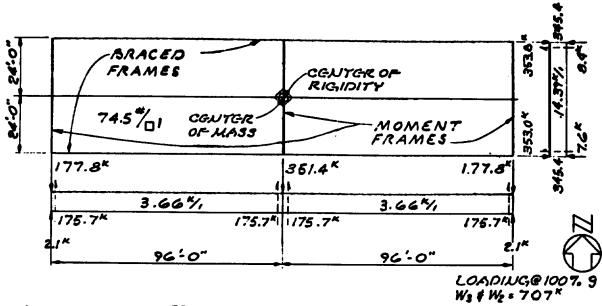
N-S LOADS

ROOF = 35.7 x 52' = 1856,4 WALLS = 41.8 %, 1898.2 %

G-W LOADS

ROOF = 35.7 × 196" = 6997.2 WALLS = 48.0*/, 7045.2*/,

Figure D-3. Continued.



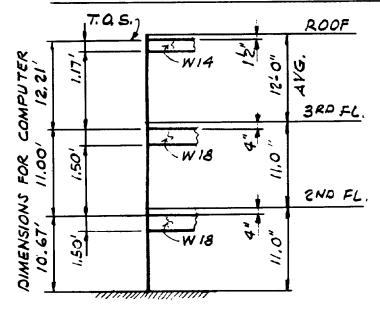
LOADS FOR 3RD FLOOR DIAPHRAGM (2ND FLOOR SAME)

FLOOR WEIGHT FOR SEISMIC = 74.5 PSF WALL WT. 5.3 PSF x 11'=58.3 4/1 N. WALL =58.3 x .75 = 44 x 192'=8448# 5. WALL =58.3 x .70 = 39.6 x /92'=7603" 83.6 % G. WALL = 58.3 x .75 = 44 x 48'= 21/2" W. WALL =58.3 x.75 = 44 x 48' = 2112# 88*/1 N-S LOADS F-LOOR = 74.5 x 48'= 3576.0 WALL = 84.0 3660.0 */ G-W LOADS FLOOR = 74.5 x 192' = 14304 WALL 88. 14392.

Figure D-3. Continued.

TRANVERSE (N-S) DIRECTION :

STEEL SPECIAL MOMENT-RESISTING FRAMES



THE COLUMN BASE IS ASSUMED FIXED.

THIS IS NOT ALWAYS
FEASIBLE, ACTUAL
FOUNDATION CONDITIONS
SHOULD BE CAREFULLY
STUDIED, AND REALISTIC
ASSUMPTIONS SHOULD
BE MADE FOR
ANALYSIS.

FRAME CHARACTERISTICS

Gravity load: The middle frame will take twice as much gravity loads as each end frame according to tributary area.

All three frames will have the same Seismic load: Assuming the roof diaphragm is flexible, and proportions. using the tributary area approach, the middle frame will take half of the seismic load at the roof level while each end frame will take one quarter. Assuming the floor diaphragms are rigid, the third floor diaphragm will distribute some of the lateral load that originates at the roof level from the middle frame to the end frames; also, because the three frames have equal stiffness, the rigid floor diaphragms will distribute one third of the load that originates at each floor to each of the three frames. The roof diaphragm is not fully flexible: the middle frame will take something less than half and the end frames something more than one quarter each of the roof load. Also the floor diaphragms are not fully rigid: the end frames will probably not get a full third of the load. The example assumes that the middle frame keeps full half of the roof load and one third of the floor loads: what is probably an excessive load from the roof tends to offset what is probably a deficient load from the floors.

Figure D-3. Continued.

TM 5-809-10/NAVFAC P-355/AFM 88-3, Chap 13

FRAME CHARACTERISTICS - cont'd

Because of accidental torsion the end frames will take some torsional forces below the third floor.

The total seismic forces being nearly the same in all frames, the design will be governed by the middle frame which takes twice as much gravity load as each end frame, and the design example will be concerned with this frame, i.e., the transverse frame on Line 4.

BUILDING PERIOD

In order to calculate lateral forces for design of the frame, the building period is needed. SEAOC provides two methods, Method A and Method B, and this example will make use of a third method, the Drift Limit Method.

Method A provides a simple formula based on the height of the building and the structural system, so it could be used as a first approximation for a preliminary design. Using $C_t=0.035$ for steel frames and $h_n=34$ feet, SEAOC eq 1-3 provides T=0.035 (34) $^{3/4}=0.49$ seconds. Method A is intentionally conservative; it tends to be a lower bound. This is particularly noticeable with steel moment frames. It would be desirable to use a longer period in order to reduce the design forces in a more realistic representation of the building; however this must be done with care because if the period is too long the preliminary design will be undersized. The code provides a limit on T by not allowing a value of C less than 80% of the value obtained by using T from Method A.

Method B is an accurate method, but it requires frame deflections which can be calculated only after a preliminary design is established.

The <u>Drift Limit Method</u> provides a period based on the assumption that the frame is so limber that it is at its maximum allowable deflection under code-prescribed loads. This provides an upper-bound period in contrast to the lower-bound period of method A. The period of a frame can be approximated by the formula $T = 2\pi\sqrt{(\delta_n/a_n)}$, where δ_n is the lateral roof displacement for the peak roof acceleration a_n . For the

BUILDING PERIOD - cont'd

equivalent static force procedure, δ_n is the roof displacement due to the prescribed forces, and a_n is approximately equal to (1.7 V/W)g, or (1.7ZIC/R_W)g, where $C = 1.25S/T^{2/3}$ and g is the acceleration due to gravity. The formula can be written

$$T=0.66 \left(\frac{\hat{\sigma}_n R_{\nu}}{ZIS}\right)^{3/4}$$
 with $\hat{\sigma}_n$ in feet.

For T > 0.7 sec., the story drift limit is 0.03/R_w. If the deflected shape is a straight line, $\delta_{\eta} = 0.030 \ h_{\eta}/R_{w}$. But it is not likely that the deflected shape will be a straight line. Let us assume that the average story drift is 0.80 times the maximum story drift; then, $\delta_{\eta} = 0.024 \ h_{\eta}/R_{w}$ and

$$T=0.040 \left(\frac{h_n}{ZIS}\right)^{3/4}$$
 for T > 0.7 sec.

For T < 0.7 sec., the story drift limit is $0.04/R_{\mbox{\tiny W}}$ and

$$T=0.050 \left(\frac{h_n}{ZIS}\right)^{3/4}$$
 for T < 0.7 sec.

The example will make use of this method. With $h_n=34$, Z=0.4, I=1.0, and S=1.5, T=0.83 sec. (T>0.7 sec.)

Period calculations, being based on framing members, are "bare-frame" periods, that is, they do not account for the participation of nonseismic frames and nonstructural elements. For <u>force</u> calculations, the calculated period will be reduced in order to account for the stiffening effects of these frames and other elements. For this example, we will divide the drift limit period of 0.83 sec. by a factor of 1.2, a number obtained by judgment.

The design lateral forces will be based on a "whole building" period of T = 0.83/1.2 = 0.69 sec.

Note that with T_A = 0.49 sec, C_A = 2.75. With T = 0.69 sec, C = 2.40 which is greater than 0.80 x 2.75 = 2.20.

FOR PRELIMINARY DESIGN FORCES LATERAL

ANALYSIS. STRESS USE T = 0.69 SEC. FOR

A-3 BUILDING:

DIRECTION: TRANSVERSE

Z = 0.4; I=1.0 Ru = 12

C = 125 s = 1.25 x 1.5 7 43 = (0.69) 7/3 = 2.40

≥

 \Im

Q W 0

T = 0.69 SEC. $F_T = 0.07 TV$

 $F_{x} = (V - F_{T}) \frac{Wh}{SWh} = 1.0 V \frac{Wh}{SWh}$ $V = \frac{ZEC}{R_{W}} W = 0.000^{-11}$

= 143 KIPS

*FT = 0 WHEN TE O.7 SEC.

= 1789 KIPS

0.045 0.080 ** ALL < 0.14) .: USF 0.14 0.087 0.103 0.133 0.133 (33) (25) FOR DIAPHRAGMS. (SEAOC I HZ;) K-FT 1826 3399 (0) W 600 $\widehat{\mathbb{S}}$ E(7) (3) x(9) V 4007M KIPS K-FT 143.0 1573 111.5 1226 30 600 20 B 61.5 31.5 1,00 143.0 50.0 Ô 0 L IJ 0.35 ઉ 1082 | 15,554 | 0.43 7,777 0.22 π_{r} 150 36,081 4×(0) Ş Ē ũ 375 1789 ¥ W \widehat{v} XIPS 375 101 1789 3 101 Ð F 44 2 $\widehat{\mathfrak{D}}$ L (2) 48 2 22 LEVEL W

Figure D-3. Continued.

DISTRIBUTION OF FORCES TO FRAMES

Since the roof diaphragm is relatively flexible, the roof forces are distributed by tributary area.

The 2nd and 3rd floor diaphragms distribute the floor forces to the frames according to their relative rigidities.

The transverse frames on lines 1, 4 and 7 are alike, and for preliminary design we may take their rigidity proportional to

$$K_1 = \frac{1/3(BASE\ SHEAR)}{DRIFT} = \frac{1/3(143)}{0.0025(34')} = 561\ kJft$$
see page 9

The longitudinal frames on lines A and C have a rigidity based on preliminary trials:

see page 2.7

$$K_A = \frac{1/2(BASE\ SHEAR)}{DRIFT} = \frac{1/2(250)}{0.30''/12} = 5000\ k/ft$$

prelim calcs (not shown)

Use Rel.
$$K_1 = 1$$
 and Rel. $K_4 = \frac{5000}{561} = 8.91$, say 9

Because of symmetry there is no "calculated" torsion. The "accidental" torsion is the story force, F, times the nominal eccentricity of 5% of the max. building dimension: perpendicular to the direction of force under consideration:

POR forces in the transverse direction.
$$M_t = F_{transv.} \times 0.05 \times 192' = 9.6 F_{transv.}$$

Torsional shear = Kd 9.6 $F_{transv.}$
 EKd^2

For forces in the longitudinal $M_t = F_{long} \times 0.05 \times 48 = 2.4 F_{long.}$

Torsional shear = Kd EKd^2

Torsional shear = EKd EKd^2 EKd^2

Figure D-3. Continued.

DISTRIBUTION OF FORCES - CONT.

			432	S=28,800			
A C	9	24 24	2/6	5184 5184	.50 FL	† 0.02 FL 7 0.02 FL	0.52 FL* 0.52 FL
			192				
1 4 7	1 1 1	96 0 96	96 0 96	9216	.33 F _T .33 F _T .33 F _T	± .03F ₇ 0 ∓ .03F ₇	.36 F _T .33 F _T .36 F _T
FRAME	REL K	d	Kd	Kde	DIRECT SHEAR	TORSIONAL SHEAR	DESIGN SHEAR

*THESE WILL BE USED FOR DESIGN OF THE LONGITUDINAL FRAMES. SEE P. 28.

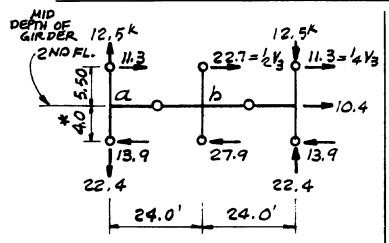
DISTRIBUTION TO TRANSVERSE FRAMES

FRAME	1	4	7
ROOF (BY TIBUTARY A	REA)	
50.0 3rd. (x.25 = 12.5 BY REL. RIGIDI		×.25 = 12.5
61.5 2ND.	× .36 = 22.1	×.33 = 20.3	×.36 = 22.1
31.5	× .36 = /1.3	x.33 = 10.4	× .36 = 11.3
143.0K	45.9K	55.7 ^k	45,9K

Figure D-3. Continued.

PRELIMINARY DESIGN

MEMBER FORCES BY PORTAL METHOD - FRAME 4



EXTERIOR COLUMN, (MOM. AT \notin GIRD.)

ABOVE d, $M = 11.3^{K} \times 5.50' = 62.2$ BELOW d, $" = 13.9^{K} \times 4.0' = 55.6$ 117.8^{K}

INTER. COLUMN (MOM. AT & GIRD)

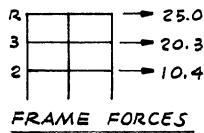
ABOVE b, M=22.7×5.50=124.9

BELOW b, "=27.9×4.0=111.6

236.5K

GIRDER (MOM. AT & COL.) $Ma = 117.8 \quad Mb = \frac{236.5}{2} = 118.3$ $V = \frac{117.8 + 118.3}{24!} = 9.84^{K}$

* ESTIMATED LOCATION OF INFLECTION CONSIDERING FIXITY OF BASE.



AT UPPER POINT OF INFLECTION, ** $M = (25.0^{k} \times 18.6^{l}) + (20.3^{k} \times 6.7^{l}) + (20.3^{k} \times 6.7^{l})$ $= 601^{k'}$

$$AXIAL = \frac{601^{K'}}{48} = 12.5^{K}$$

AT LOWER P. L.

$$M = 601 + (45.3 \times 9.5) + (10.4 \times 4.0) = 1073K'$$

$$AXIAL = \frac{1073}{48} = 22.4^{k}$$

**18.6' IS APPROXIMATE PISTANCE FROM ROOF TO POINT OF INFLECTION.
6.6' IS APPROXIMATE DISTANCE FROM TOP OF 3PP FLOOR SLAPS TO POINT OF INFLECTION.

Figure D-3. Continued.

PRELIMINARY DESIGN - CONT. FRAME 4

INTERIOR COLUMN

CL. VERTICAL LOAD ON CENTER FRAME

ROOF DL: 0.0257 KSF x 24'x32' = 19.8

3RP FLR. DL+LL = (0.0745 + 0.021) x 24 x32' = 73.3

2NP FLR. RED. LL = 73.3

W=166.4 K

BY SYMMETRY, M = 0b. SEISMIC LOAD, FROM P. 13 P = 0 K-ft $M = 27.9^K \times (4.00-0.75) = 90.7$ AT FACE OF GIRDER

C. VERTICAL + SEISMIC $P = 166 + 0 = 166^K$ $M = 0 + 90.7 = 90.7^K$ USE AISC 9TH EDITION, P. 3-9 $TRY W 14 \times 61$, P. 3-24 Bx = 0.194

Pequiv. = $\frac{166^K + 0.194(90.7 \times 12\%)}{1.33} = 284^K$ FOR $h' = 9.5' \frac{W14 \times 61}{T = 6.40}$ ALLOWS 334^K

EXTERIOR COLUMN

VERTICAL + SEISMIC $P \cong IGG/2 + 22.4 = IO5^{K}$ $M \cong 50^{K-1}(est) + 90.7/2 = 95^{K-1}$ $P \in QUIY. = 24G$ $\frac{W14 \times 46}{I = 485}$ ALLOWS 24G^K

USE 14" COLUMNS FOR CONTROL OF DEFLECTIONS AND USE THE SAME SECTION FULL HEIGHT.

PRELIM. DESIGN - CONT.

FRAME 4

GIRDER- ZND FLOOR

VERTICAL LOAD AT CENTER COLUMN

$$W_{0+L} = (0.0746 + 0.021) \times 32' = 2.38 + 0.67 = 3.05 \% / 1$$

$$W_{O+L} = 3.05 \times 24' = 73.2^K$$

$$M \approx \frac{WL}{l^2} = \frac{73.2 \times 24}{l^2} = 146.4$$

$$SEISMIC$$
 $M = 118.3$

Figure D-3. Continued.

PRELIM. DESIGN - CONT.

WD = 0.82 x 24'= /9.7K

FRAME 4

GIRDER - 3RD FLOOR

VERTICAL LOAD - SAME AS 2ND;
$$M = |46^{K^{\dagger}}]$$

SEISMIC $G:25^{K}$ TO EXT. COL.

 $V_{R} = 25.0$ $I2.5^{K}$ TO INT. COL.

3RD FLR. $I2.5^{K}$ $G:25^{K}$ G

SE/SM/C 5 12.5 - GIRDER M = 12.5×G.04 2 37.8

VERT. + SEISMIC = M = 39.4 + 37.8 = 77.2 ÷ 1.33 = 58.0 KI

WIF x 22 ALLOW 58 OK FOR STRESS

USE WIF x 30 ALLOW 83 USE WIDER FLANGE
FOR BETTER DETAILS

Figure D-3. Continued.

CRITERIA FOR FINAL DESIGN - FRAME 4

1. BUILDING PERIOD

- a. For calculating frame design forces, use the whole building period estimated as T = 0.69 sec. (p. 9)
- b. For calculating drift, use the bare frame period. This will be obtained from the computer analysis or from the use of Method B. Method B will give a frame period of 0.83 sec.

2. DESIGN FORCES

As indicated item 1a above, use the building period of T = 0.69 sec. and the associated base shear of 143^k and frame shear of 55.7^k (p. 12). This is the input for the computer analysis (p. 18).

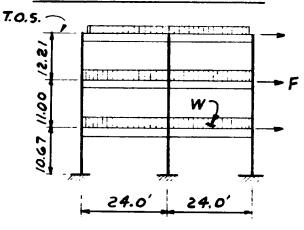
3. <u>DEFLECTIONS</u>

The deflections obtained from this analysis, based on whole building forces, will be modified for the calculation of drift under bare-frame forces (p.20).

FRAME ANALYSIS - FRAME 4

COMPUTER INPUT

KIPS, FEET, SECONDS



RIGID FRAME. STEEL: E = 4,176,000 KSF COLUMN BASES FIXED.

	EXT. COL.	INT. COL.
SIZE	W/4×48	W/4x 61
I	0.0234	0.0309
A	0.0979	0.1243
Aw	0.0325	0.0365

LEVEL			GIRI	DER			W = V 187 ^K 5 233 7.233 7.	MASS LATERA	LATERAL
	SIZE	I	A	Aw	WAL	WLL		= W/9	FORCE
R	W/4×30	0.0140	0.06/3	0.0260	0.82	0.64	187K	5.81	25.0
3	W/8×55	0.0430	0.1125	0.0491	2,38	0.67	233	7.24	20.3
S	W/8×60	0.0476	0.1229	0.0527	2.38	0.67	2 3 3	7.24	10.4
DAT	A FRON	PAGE	ļ	1	14	£ 15	566		11

TRIB. ROOF WT. 18 2 OF TOTAL SINCE DIAPH. 15 FLEXIBLE TRIB. FLOOR WT. 15 0.33xTOTAL ACCORDING TO REL. RIG. OF FRAMES. 9= 32.2 FT./SEC.

Figure D-3. Continued.

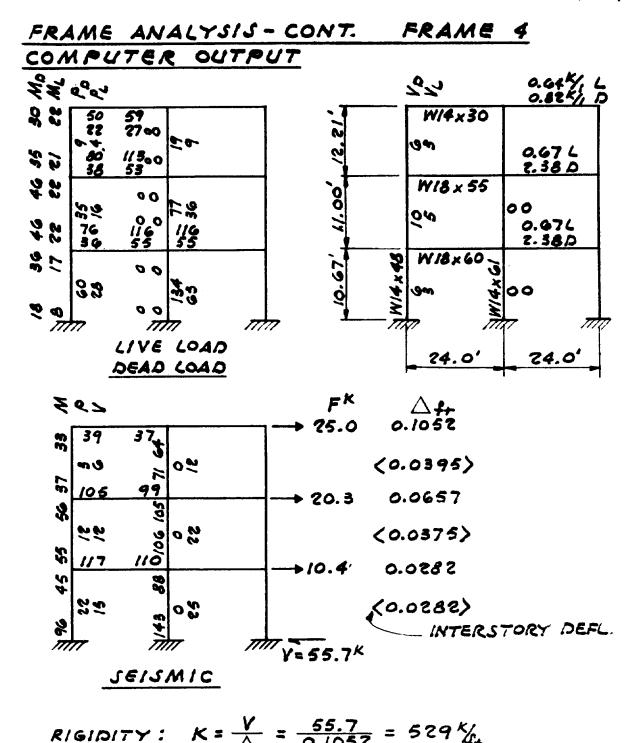


Figure D-3. Continued.

TM 5-809-10/NAVFAC P-355/AFM 88-3, Chap 13

FINAL DESIGN - CONT. FRAME 4 - DRIFT

Before proceeding with the detailed final design, we will check the drift.

From the computer analysis, the maximum drift is 0.0395 ft. (p. 19). This is based on the whole building period T = 0.69 sec. and the frame shear of 143^k (p. 10).

For the drift check, use the deflection associated with the bare frame period T = 0.93 sec. (p. ?).

C =
$$(1.25 \times 1.5)/(0.73)^{2/3} = 2./2$$

CS = ZIC/R_u = $(0.4 \times 1.0 \times 2./2)/12 = 0.071$

Base Shear = $0.07! \times 1789 = /27^{k}$

Multiply deflections from frame analysis by the ratio 12.7/143 = 0.89.

Maximum drift = $0.29 \times 0.0395' = 0.935'$

Allowed drift = (0.03/12)x12' = 0.030' (T = 0.23 > 0.7)

The frame drift is 17% over the limit. It should be stiffened. There are three options:

- (1) increase the member sizes
- (2) use more than three frames
- (3) make the roof diaphragm rigid

We will change the interior column to W14x68 and proceed with the detailed design. Further changes may be necessary, and we will make a final drift check after other checks are completed.

Note that the assumed condition of fixed columns bases is difficult to achieve. If the bases are not fully fixed, the frame will be more flexible than assumed, and the frames would have to be further stiffened.

FINAL DESIGN - CONT. FRAME 4 MEMBER STRESSES

(1) SAMPLE CALCULATION FOR 2ND FLR. GIRDER

	Mp	ML	ME	MAL	MD+L+E 1.33
AT EXT. COL.	76 116	3 G 5 5	117	112	772

DES. M = 211K' UNBRACED LENGTH = G' W18xGO ALLOW M = 21GK' AISC 9TH ED., P. 2-166

PROVIDE GIRDER BRACING PER SEAOC 4F8:

MAX UNBRACED LENGTH = 96ry

= 76 (1.69) = 13.5 PT. > 6FT.

12

BRACING IS ADEQUATE

(2) SAMPLE CALCULATION FOR COLUMN, SEE NEXT PAGE.

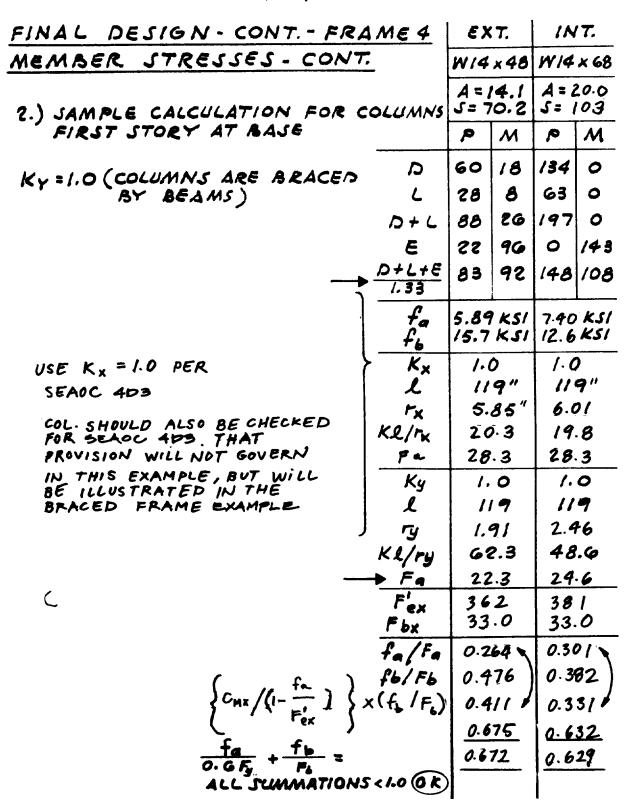


Figure D-3. Continued.

GIRDER-COLUMN CONNECTION

Girder data W18x60 Gr. 36

Girder strength in flexure

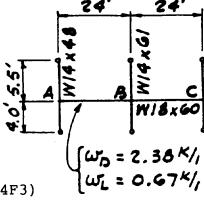
This is $M_s = ZF_y$ (SEAOC 4C2). This is the same as plastic moment M_p given in the in the AISC Plastic Design Selection Table:

$$M_p = 369 \text{ k}'$$

Girder stability W18x60

$$b/2t_f = 5.44 < 52/\sqrt{36} = 8.66 \text{ (SEAOC 4F3)}$$

 $d/t_w = 43.5 \le 412/\sqrt{36} = 68.7 \text{ (AISC Ch.N)}$



Requirements for girder-column connection

SEAOC 4F1a requires development of the lesser of the strength of the girder in flexure (M_s) and the moment associated with the panel zone shear strength. The requirement of this manual is to develop the strength of the girder in flexure. This is accomplished according to SEAOC 4F1b.

Girder flange connection to column

Provide full butt-weld connections of the flanges to the columns. (SEAOC 4F1b(1))

Girder web connection to column

Design shall be based on the gravity loads plus the seismic load associated with compliance with SEAOC 4F1a. (SEAOC 4F1b(2))

Vert. shear:
$$V_G = (2.38+0.67)24'/2 + (171-112)/24 = 39.1$$

Seismic shear $= 2M_s/L = 2x369k'/22.83' = 32.3$
Design V $= 71.4k$

The method of developing the flexural capacity of the web depends on the relative size of the flanges, i.e., whether the flange strength (bt_f(d-t_f) x F_{y)} is greater or less than 70% of the total strength (0.7 Z_xF_y)

Figure D-3. Continued.

TM 5-809-10/NAVFAC P-355/AFM 88-3, Chap 13

GIRDER-COLUMN CONNECTION

Girder web connection to column - cont.

$$b = 7.555$$
"; $t_f = 0.695$ "; $d = 18.24$ "; $Z_x = 123in^3$

For 70% of entire section:

$$0.7 Z_x = 0.7 \times 123 = 86in^3$$

For the flanges alone:
$$bt_f(d-t_f) = (7.555")(0.695")(18.24-0.695") = 92.12 in^3$$

As 92.12 > 86, the connection can be made by welding and/or high-strength bolting according to (SEAONC 4F1b(2) (a)

If the chosen girder had had flange strength less than 70% of the total strength, the conventional web connection would have had to be supplemented with additional welding (to the web at the top and bottom of the shear tab on the column) according to SEAOC 4F1b(2) (b).

Girder web connection design

Assume 1" A325-SC bolts are selected,

Use 4 bolts: Bolt Strength,
$$V = 4 \times 1.7 \times 13.7$$

= 93.2k > 71.4

Shear plate:

$$Z = 0.3125 (12.5)^{2}/4 = 12.2 in^{3}$$

 $f = 71.4k \times 2''/12.2 = 11.7 ksi < 36$
 $v = 71.4/(0.3125 \times 12.5) = 18.3 ksi < 0.55 \times 36 = 19.8$

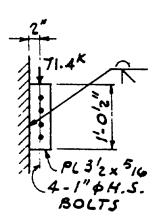


Figure D-3. Continued.

GIRDER-COLUMN CONNECTION

Girder-column panel zone -- Sample calc. for Joint B

Panel zone strength (SEAOC 4F2)

This is the moment corresponding to the development of the panel zone shear strength. This moment shall be the moment due to gravity loads plus 1.85 times seismic loads but need not exceed 0.8 times the summation of M at the beams.

Calculate moment arm between girder flanges: $d-t_f = 18.24-0.69 = 17.55$ "

Panel zone shear:

Gravity + 1.85 x seismic

One side of column, with D + L: $M_{D+L} = 171$; 1.85 $M_E = 1.85 \times 110 = 204$

Other side, with D: $M_D = -116$; 1.85 $M_F = 204$

Sum of girder moments = 171 + 204 - 116 + 204 = 463

0.8 M

Sum of girder moments = $2 \times 0.8 \times 369 = 590 \text{ k}$

Use panel zone strength associated with 463 k'.

(See SEAOC 4F10 for Drift calculations.)

Shear

One side, top flange force = $(171+204)(12)/17.55^{"}$ = 256 K Other side, top flange force = $(-116+204)(12)/17.55^{"}$ = 60 K

Column shear above joint:

sum girder moments = 463column height = 4.0 + 5.5 = 9.5' column shear = 463 / 9.5 = 49 k

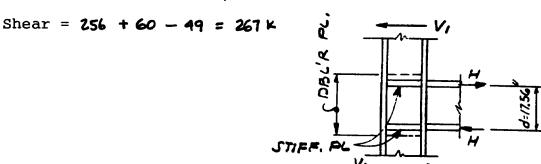


Figure D-3. Continued.

TM 5-809-10/NAVFAC P-355/AFM 88-3, Chap 13

GIRDER-COLUMN CONNECTION -- Joint B -- cont.

Panel zone thickness

For thickness, t', the panel zone can develop

$$V_i = 0.55 F_v dt'$$
 (modified SEAOC formula 4-1)

where t' is the effective thickness which consists of the combined thickness, t, of the web and doubler plate modified by the contribution of the the columns flanges (see below)

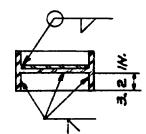
For W14 x 68 Grade 50 column
$$V_j = 0.55 (50)(14.04)t' = 386t' \text{ kips}$$

For joint shear of 235 k, req'd t' =
$$267/386 = 0.69$$

t' = t [1 +
$$3b_c t_c^2/d_b d_c t$$
]
= t [1 + $3(10.03)(0.72)^2/(18.24)(14.04)t$]
= t [1 + $0.061/t$] = t + 0.061

$$0.69'' = t + 0.06$$

 $t = 0.63''$



For column web thickness of 7/16" req'd thickness of doubler plate = 0.63" - 7/16 = 0.19"

Use a 1/4" minimum Grade 50 doubler plate, or consider using a column with a thicker web.

Check SEAOC 4F2b:

$$d_z = d - 2t_f = 18.24 - 2 \times 0.695 = 18.10$$
 (from girder)
 $w_z = 14.04 - 2 \times 0.720 = 12.50$ (from column)
 $(d_z + w_z) / 90 = 0.34 < 0.63^{\circ}$ OK

Continuity plates

Provide continuity plates per SEAOC 4F2c